Virtual Testing Environment for Flexible Drone Flight Simulation and Design Analysis

Research Question

- What improvements can be made to the data collection process for flexible drones with the use of testing in a virtual environment?
- What effect does using a virtual environment have on the testing of flight pathing and execution of a drone's hardware and software?

Research Progress

- Got set up with the software needed • Further improve the accuracy of the cad model until the for the project • Tested different methods/programs model is close to 100% for modeling and data collection accurate
- Created a 3d cad model prototype for the drone to be simulated
 - made of both modified existing models and custom made ones
- Implement the drone fully • Implemented most parts- custom or modified- into the simulation into the simulation to allow individually- not combined into one for flight tests and not just physical analysis drone yet

Modeling and Data Analysis of Parts

Exploded View of Prototype Drone Model



Modeling and cross-referencing





Incorporating into the Simulation







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Future Plans

• Compare the manual computations & model analysis data to improve computer analysis methods

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NF0 NF0 NF0 NF0 NF0 NF0 NF0 NF0 NF0 NF0	<pre>[navigator] RTL: climb to 490 m (2 m above destination) [commander] Failsafe mode deactivated [navigator] RTL: return at 490 m (2 m above destination) [navigator] RTL: land at destination [commander] Linding detected [commander] Disarmed by landing commander] Disarmed by landing commander] Armed by internal command [navigator] Using minimum takeoff altitude: 2.50 m [logger] Start file log (type: full) [logger] [logger] ./log/2022-01-28/19_21_57.ulg [logger] closed logfile, bytes written: 949021 [logger] Opened full log file: ./log/2022-01-28/19_21_57.ulg [commander] Failsafe enabled: No manual control stick input [commander] Failsafe mode activated [tone_alarm] battery warning (fast) [navigator] RTL: landing at home position. [navigator] RTL: climb to 490 m (2 m above destination) [commander] Failsafe mode deactivated [navigator] RTL: climb to 490 m (2 m above destination) [commander] RTL: landing at destination</pre>	

Data comparison: Manual computation vs Computer Analysis

Equations (for manual calculations)

	1 V
J_{xx}	$\frac{2}{5}MR^2 + M(cg_y^2 + cg_z^2) + 2m(cg_y^2 + cg_z^2)$
J_{yy}	$\frac{2}{5}MR^{2} + M(cg_{x}^{2} + cg_{z}^{2}) + m((l - cg_{x})^{2})$ $m((-l\cos\beta_{2} - cg_{x})^{2} + (h - cg_{z})^{2})$
	$m((1000p_2 \circ g_x) \circ (n \circ g_z))$
J_{zz}	$\frac{\frac{2}{5}MR^2 + M(cg_x^2 + cg_y^2) + m(cg_y^2 + (l - d_y^2))}{(l \sin \beta_2 - cg_y^2)^2 + (l - d_y^2)^2 + (l - d_y^2)^2}$
	$+m((i \sin p_2 - cg_y) + (-i \cos p_2 - cg_x))$
$J_{xy} = J_{yx}$	$Mcg_ycg_x - mcg_y(l - cg_x) - mcg_y(-l -$
$J_{yz} = J_{zy}$	$Mcg_ycg_z + 2mcg_ycg_z + m(-l\sin\beta_1 - c_y)$
$J_{zx} = J_{xz}$	$Mcg_xcg_z - m(l - cg_x)cg_z - m(-l - cg_x)$
cg _{xvz}	$(-Cl(sin(B_1)-sin(B_2))), -Cl(cos(B_1)+c)$

Note:

The Machine calculation is a black box, it does not explicitly follow the formulas listed above

and Pamir on Grabcad respectively (IROS), 2020, pp. 1364-1370, doi: 10.1109/IROS45743.2020.9341730.

Drone Solidworks Model



[a] Custom model for the GemFan Hurricane, modeled using Solidworks 2021, using official specifications, and a modified PolyCarb material file from Solidworks, custom made for greater accuracy than publicly available models could provide.

[b] Rigid Arm for the Squeeze drone propellor arms, ported from a .stl mesh file using Solidworks 2021, using a custom material for PLA plastic, edited for software performance in Solidworks 2021.

[c] Adafruit BNO055, made in Solidworks 2021 by Pamir on GrabCad, resized using Solidworks 2021, custom metal material made in solidworks for density and material properties accuracy.

 $((l\sin\beta_1 - cg_y)^2 + (h - cg_z)^2) + m((l\sin\beta_2 - cg_y)^2 + (h - cg_z)^2)$ $(-l - cg_z^2) + m((-l - cg_x)^2 + cg_z^2) + m((-l \cos \beta_1 - cg_x)^2 + (h - cg_z)^2)$

 $(cg_x)^2$) + $m(cg_y^2 + (-l - cg_x)^2) + m((-l\sin\beta_1 - cg_y)^2 + (-l\cos\beta_1 - cg_x)^2)$

 $(-l\sin\beta_1 - cg_y)(-l\cos\beta_1 - cg_x)$

 $(k - cg_y)(h - cg_z) + m(l\sin\beta_2 - cg_y)(h - cg_z)$

 $)cg_z + m(-l\cos\beta_1 - cg_x)(h - cg_z) + m(-l\cos\beta_2 - cg_x)(h - cg_z)$

B₁,B₂ 0 Degrees 95*10⁻³Kg 710*10⁻³Kg Μ -3*10⁻²m 5*10⁻²m 12.5*10⁻²m 8.71*10⁻²

+cos(B₂)), 2Ch)

Results of calculations

Method of calculating	Results (in Kg * m²) J _{xx} ,J _{yy} ,J _{zz} ,J _{xy} ,J _{zy} ,J _{xz}
Manual Calculation	0.0037,0.0037,0.0067,0.0000,0.0000,0.0000
Machine calculation	0.0041,0.0041,0.0077,0.0000,0.0000,0.0000

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[1] K. Patnaik, S. Mishra, S. M. R. Sorkhabadi and W. Zhang, "Design and Control of SQUEEZE: A Spring-augmented QUadrotor for intEractions with the Environment to squeeZE-and-fly," 2020 IEEE/RSJ International Conference on Intelligent Robots and Systems



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