

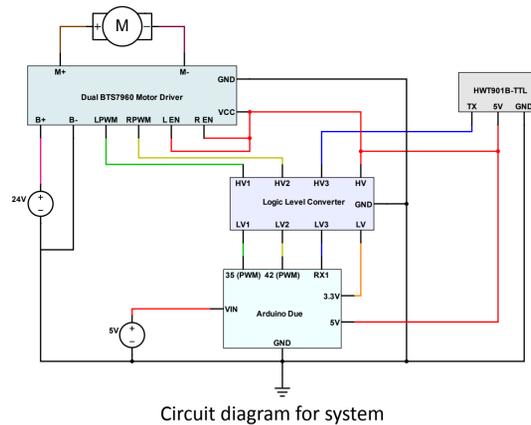
# Design, Modeling, And Control of a Single-Axis Self-Balancing Platform

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## System Description



Circuit diagram for system

Disturbance mass: 200g  
Platform mass: 558 g  
Platform length: 12 in



Platform and disturbance mass

## Plant:

Transfer Function from voltage  $u$  to angle  $\theta$   
Characterized by motor current-to-torque, armature voltage, and load torque relationships, as well as  $\theta = \dot{\omega}$   
Armature inductance and dynamic friction approximated as zero  
Parameters found from motor performance curves

$$\frac{\theta}{u} = \frac{\frac{K_t}{R_a}}{Is^2 + K_e \frac{K_t}{R_a} s}$$

$K_t$  is the motor torque constant

$K_e$  is the motor back-EMF constant

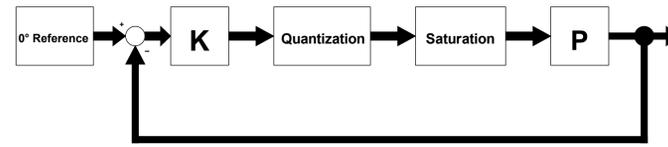
$R_a$  is the armature resistance

$I$  is the platform (unloaded) moment of inertia

## Sensor considerations:

Converting angle reading from upside down sensor to true angle  
UART interference due to computer serial port  
Allowing time for serial data to clean out before initializing  
Low sensor noise, but implementation of LPF anyway

## Controller Design



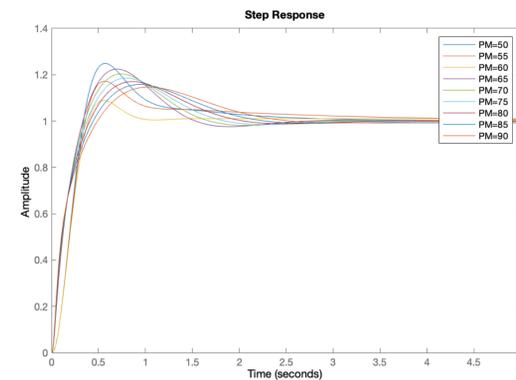
Block diagram for control system

## Controller Design Methods:

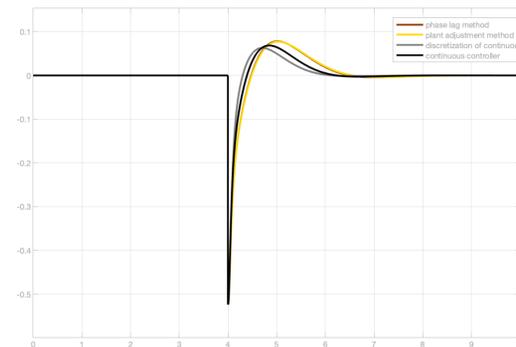
Design CT controller for BW and PM; discretize with ZOH  
Discretize plant with ZOH, convert to CT with Tustin, design CT controller, then discretize with ZOH.  
Design CT controller for BW and PM with ZOH lag accounted for in PM-based calculations  
Design CT controller for pole placement & step response, discretize with ZOH

## Design for Bandwidth and Phase Margin: PI(D):

Choose a bandwidth and create a family of controllers for a range of phase margins  
Some require two poles to generate sufficient phase. These are PID.  
Two-pole controllers have more aggressive initial response, which is ideal for preventing an object from falling



Family of step responses for continuous PI(D) controllers with bandwidth  $\approx 5$  rad/s



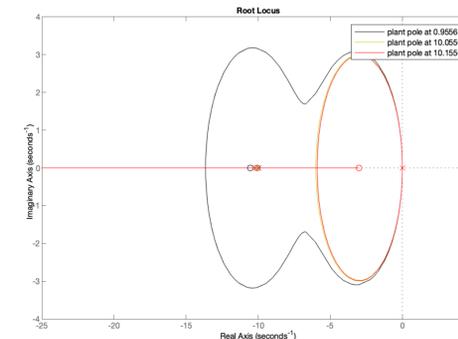
Comparison of disturbance response for different BW/PM methods (BW = 5; PM = 90°)

## Controller:

Discrete Controller  
Low-Pass Filter ( $\omega_g = 100$ )  
Output Quantized to  $\pm 10000$  steps due to PWM  
Output Limited to  $\pm 24V$   
No prefilter needed because of unchanging setpoint (controller designed for disturbance rejection)

## Pole Cancellation Considerations:

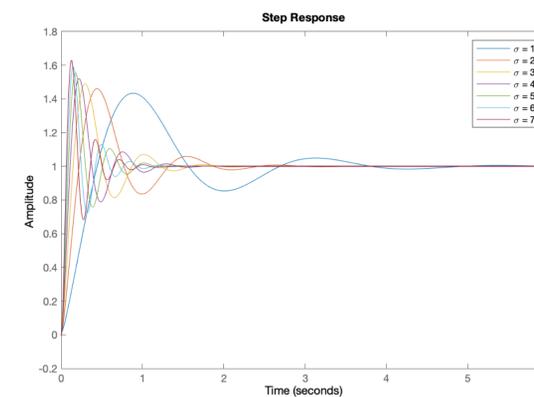
Little danger in overestimating pole  
Underestimating pole could lead to deformations in root locus  
These deformations are less significant for smaller polar angle and gain



Root locus plot describing effects of cancelling an incorrectly modeled pole

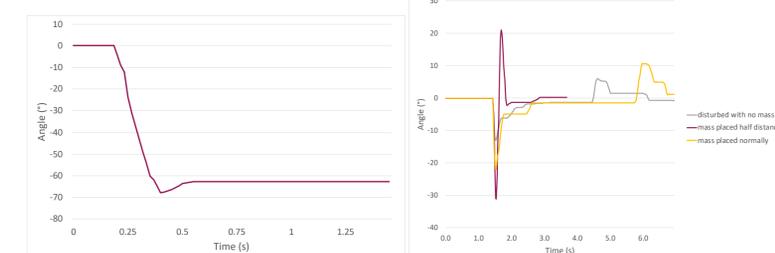
## Design for Pole Placement:

One zero at modeled plant fast pole  
Choose a polar angle and create a family of controllers for a range of sigmas  
 $\theta$  corresponds with overshoot,  $\sigma$  is inversely proportional to settling time  
If implementing, discretize with ZOH  
Overshoots differ because of the inclusion of the LPF



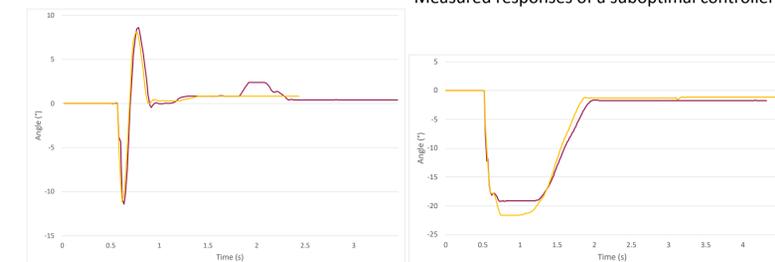
Family of step responses for continuous PI(D) controllers with  $\theta \approx 20^\circ$

## Results



Response to mass when motor is unpowered

Measured responses of a suboptimal controller



Response with controller  $\theta = 20^\circ, \sigma = 7$

Response with controller BW = 5, PM = 90°

## Observations:

For the purpose of catching a falling mass, a more aggressive transient response is desirable. This can't be increased infinitely because of the limit to voltage supplied.  
Controllers designed for BW and PM did not respond similarly to their simulations, but had high enough stability margins to still function

## Nonlinearities:

While there was little dynamic friction, static friction is significant  
With the mass on the end, it would take 5–6 volts for movement to begin  
Makes it difficult for the system to settle into a steady state gently  
Torque due to addition of mass  
Controllers were able to stabilize for this, but not able to design for this

## Next Steps:

Apply knowledge of software handling of this particular sensor to the controlled ascent of a rocket  
Apply knowledge of software handling of discrete control system (state recording, interrupt timing)  
Use a different motor or find ways to account for static friction in rocket controller design  
Actuating fins will likely be a lower torque system, especially due to not having to work against gravity, but aerodynamic forces will still cause torques  
System will need to be much faster, and overshoot may be less desired